



OpenCAEシンポジウム2011 ESIによるOpenFOAM 事例

2011/12/2

日本ESI 株式会社
三邊 考志

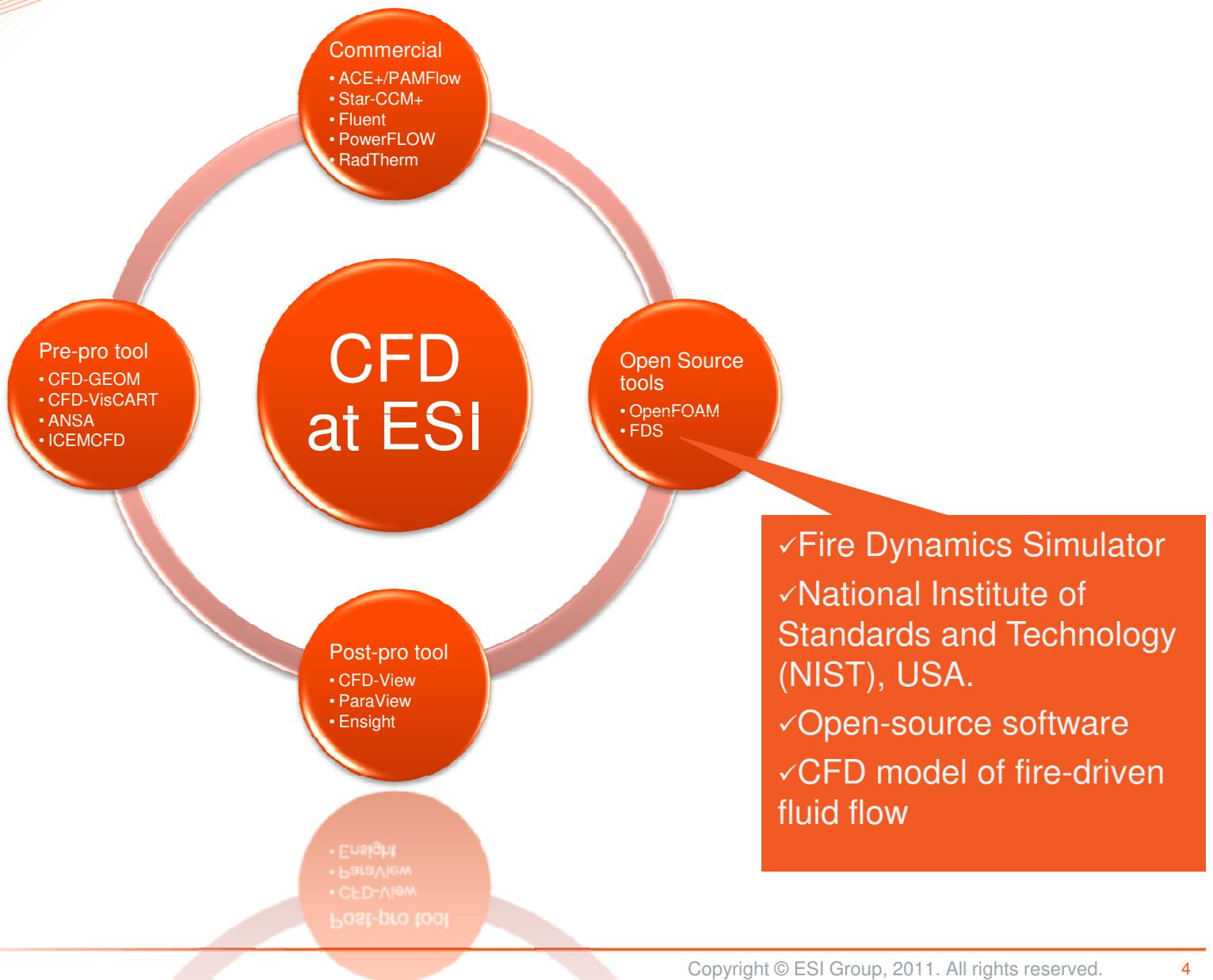
- ESI CFDのご紹介
- 背景
- 事例紹介
 - 商用CFDコードとの比較
 - SanpyHexa vs. VisCART
 - 最適化ツールとの連携事例

- 1973 航空宇宙、原子力産業のコンサルタント会社として設立
- 1985 PAM-CRASHを開発
- 1997 Framatome(SYSTUS, SYSEWLD)を買収
- 2000/7/6 Nouveau Marché に上場
- 2000-2003
 - STRACO, VASCI, ProCASTを獲得
- 2004 CFDRCよりソフトウェアプロダクト部門(CFD-ace)を獲得
- 2006
 - IPS International から人体モデルの部門を買収
 - 中国のAET Technology Internationalを統合
- 2008-2009
 - Vdot (PLM ツール) を獲得
 - イタリアとブラジルに子会社を設立
 - CFD /CAEコンサルティング会社 Mindware Engineering を買収



- 12 subsidiaries
- 30カ国以上
- 15の支社
- 従業員数 約780

ESI CFD Experience



- CFDに掛るコストの削減
- 最もボリュームのあるアプリケーション(設計者の範疇)には汎用ソフトは高機能すぎる
- 汎用ソフト価格の変化
 - 商用コードを切り替える?
 - OpenFOAMでも良いのでは?

商用コードとの比較事例

External Aerodynamics

事例(コード比較): External Aerodynamics (Incompressible Turbulent Flow: Sample Car)

- Comparison is made between the results of OpenFOAM, Fluent and Star-CCM+.

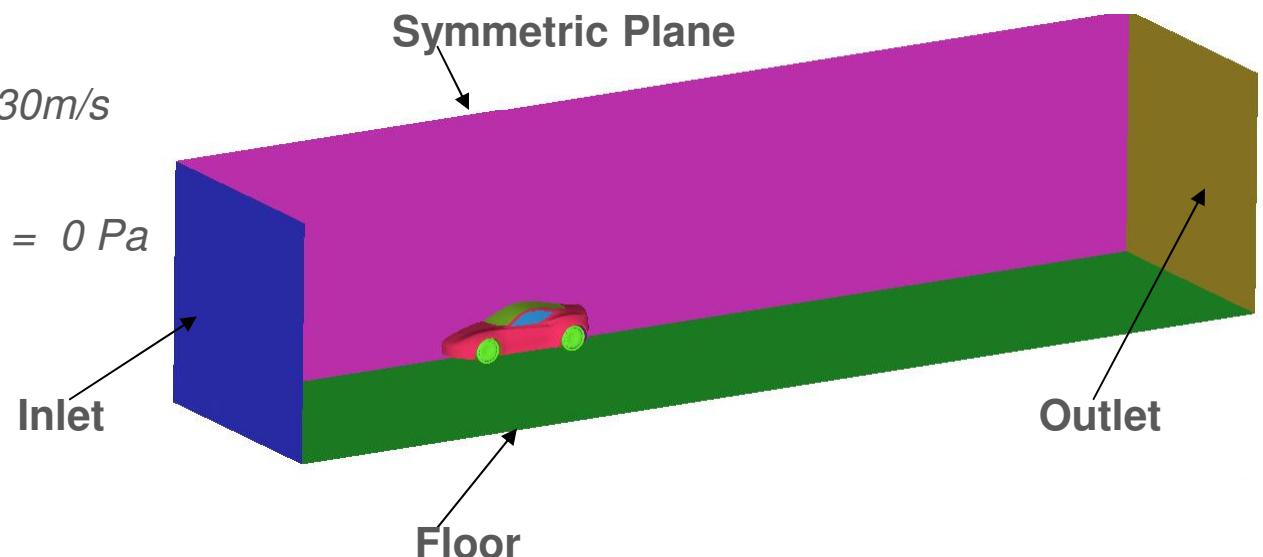
- Boundary Conditions

- Inflow:

- $\text{Velocity Inlet} = 30 \text{ m/s}$

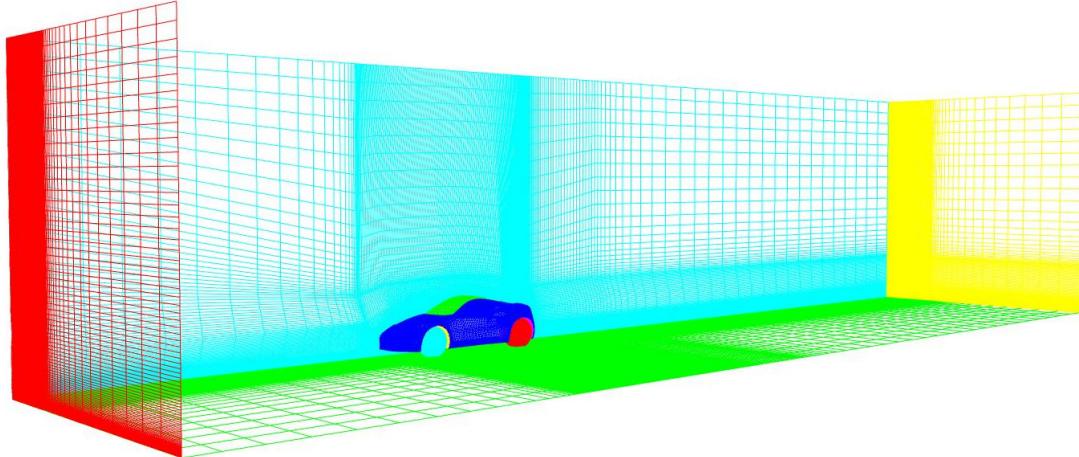
- Outflow:

- $\text{pressure outlet} = 0 \text{ Pa}$

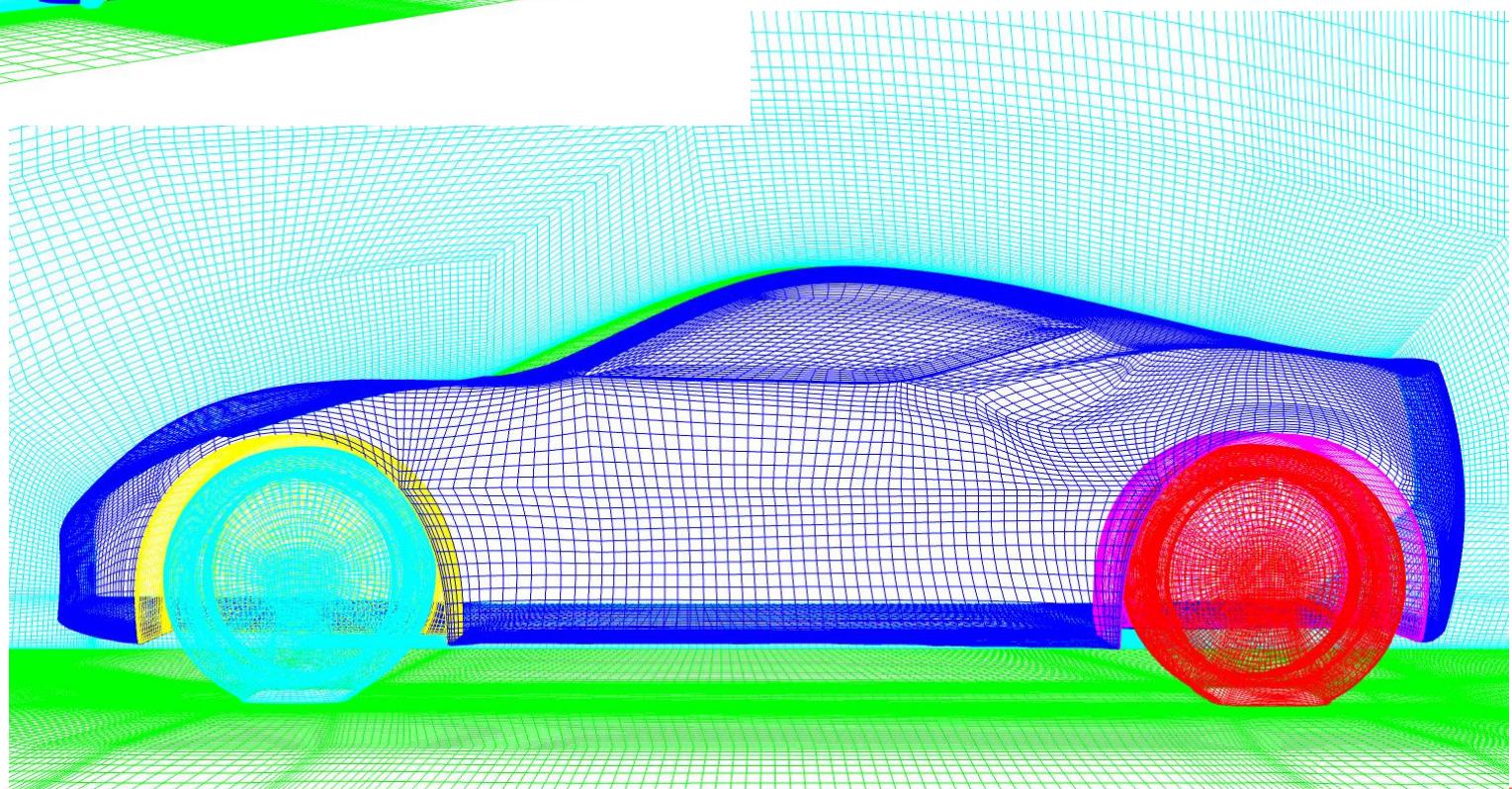


- OpenFOAM Solver: simpleFoam

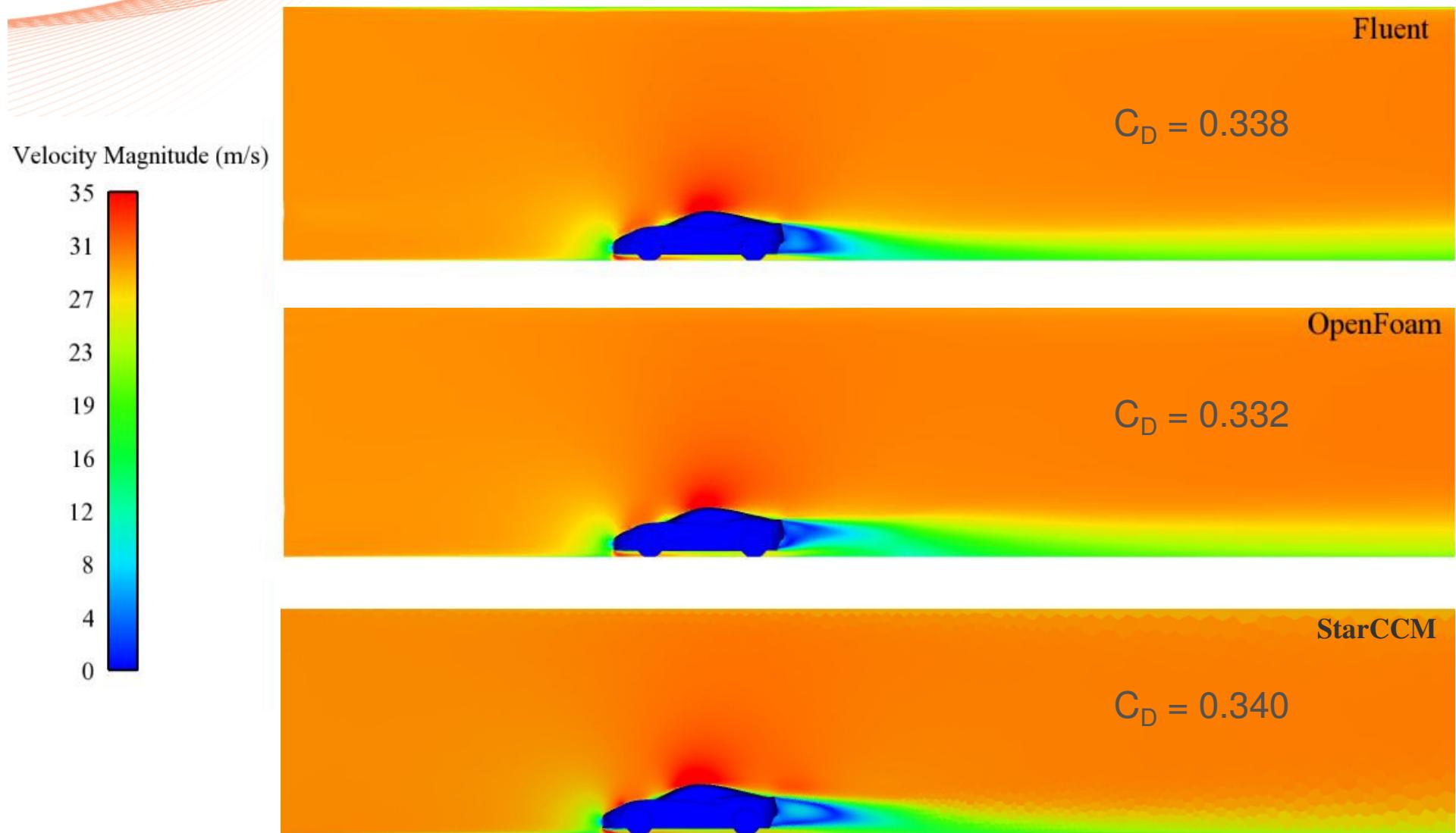
事例(コード比較): External Aerodynamics (Mesh Details)



Hexa Mesh = 2.8 Millions



事例(コード比較) : External Aerodynamics (Velocity Distribution at Symmetry Plane)



事例(コード比較): External Aerodynamics (Static Pressure Distribution at Symmetry Plane)

Static Pressure (Pa)

150

78

6

-67

-139

-211

-283

-356

-428

-490

-500

Fluent

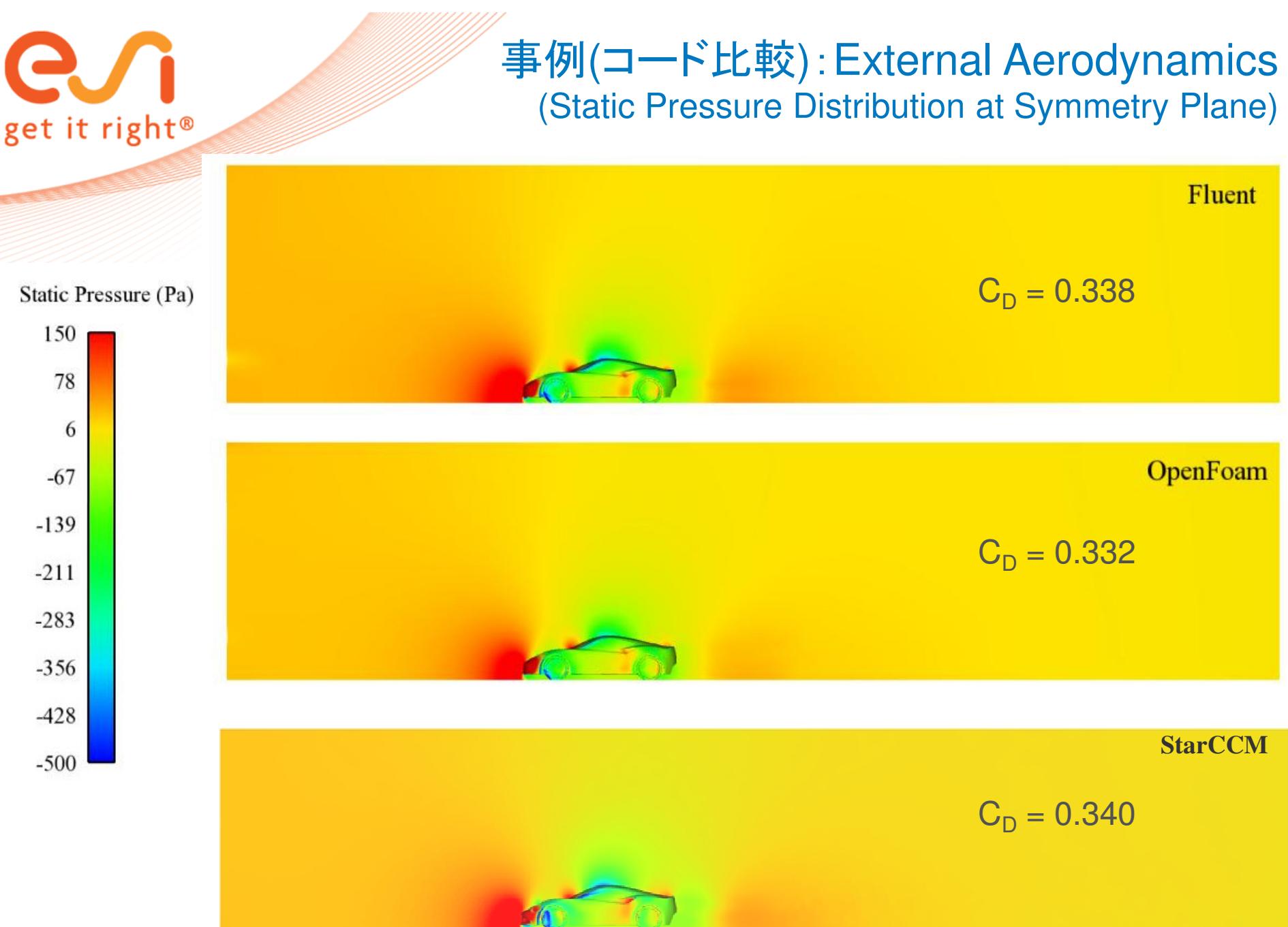
$$C_D = 0.338$$

OpenFoam

$$C_D = 0.332$$

StarCCM

$$C_D = 0.340$$



商用コードとの比較事例

実形状データでの比較

External Aerodynamics

- **Objective:** To Analyze the different effects caused by the air dam on the following attributes:

1. Drag
2. Frontend cooling
3. Comparisons with another commercial code and Test data

- **Boundary Conditions:**

- Inflow:
 - *Velocity Inlet = 50km/hr and 110km/hr*
 - Outflow:
 - *pressure outlet = 0 Pa*
 - Fan at 12 Volt
 - Heat Exchangers

- **OpenFOAM Solver:** porousSimpleFoam (1.5)

配布資料掲載
不可

事例(コード比較) :Aerodynamics
Two cases: With and Without Airdam



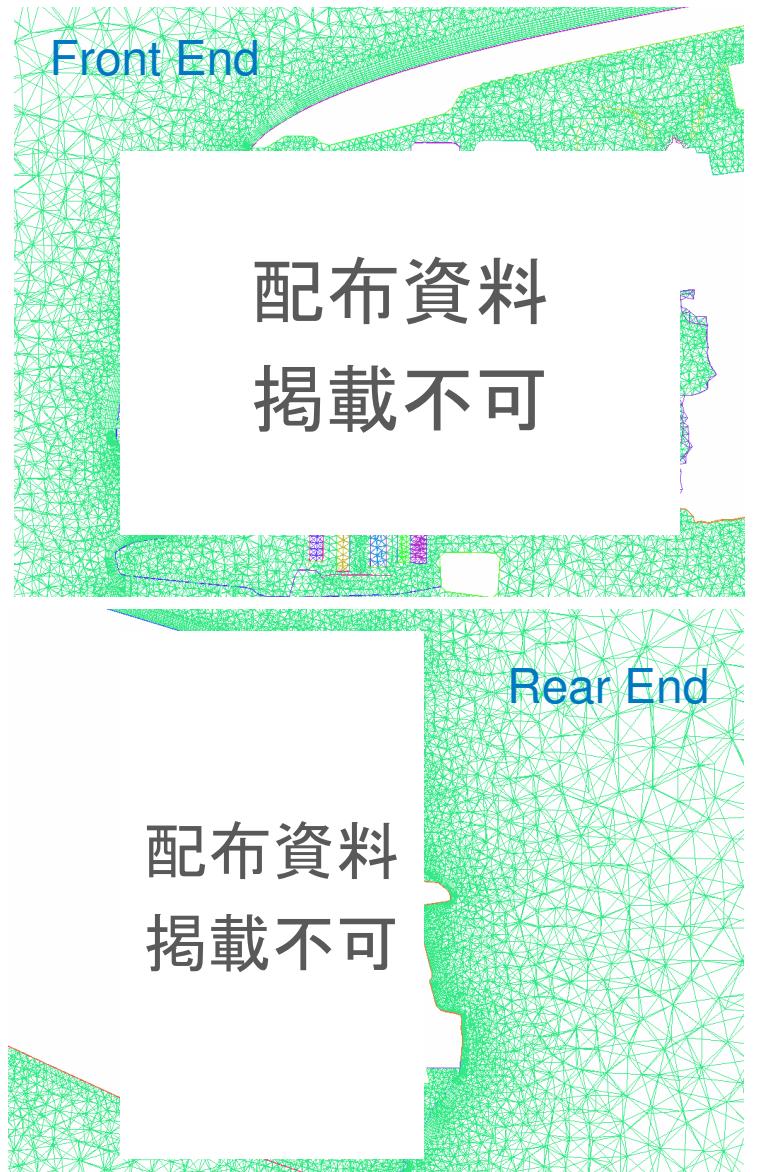
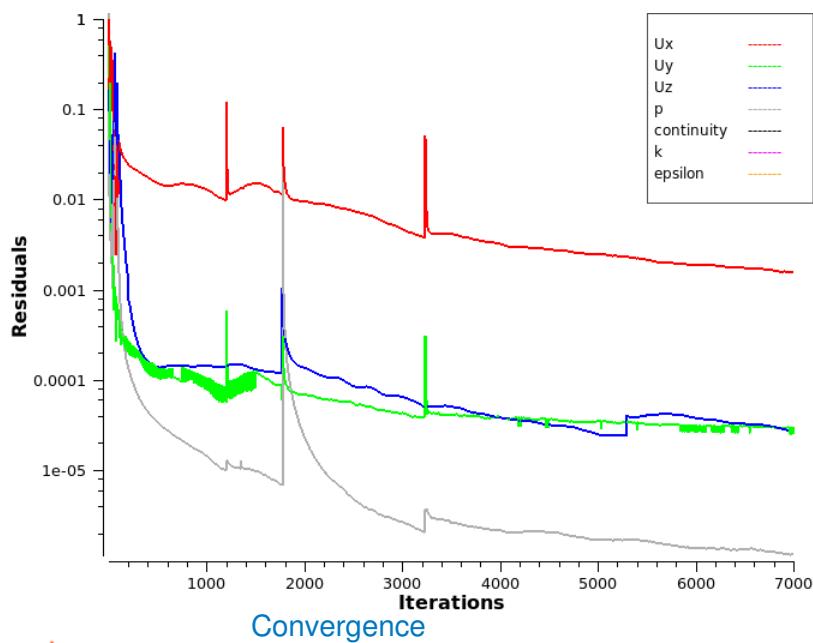
With Airdam



Without Airdam

事例(コード比較) :Aerodynamic (Incompressible Turbulent Flow: Real Car)

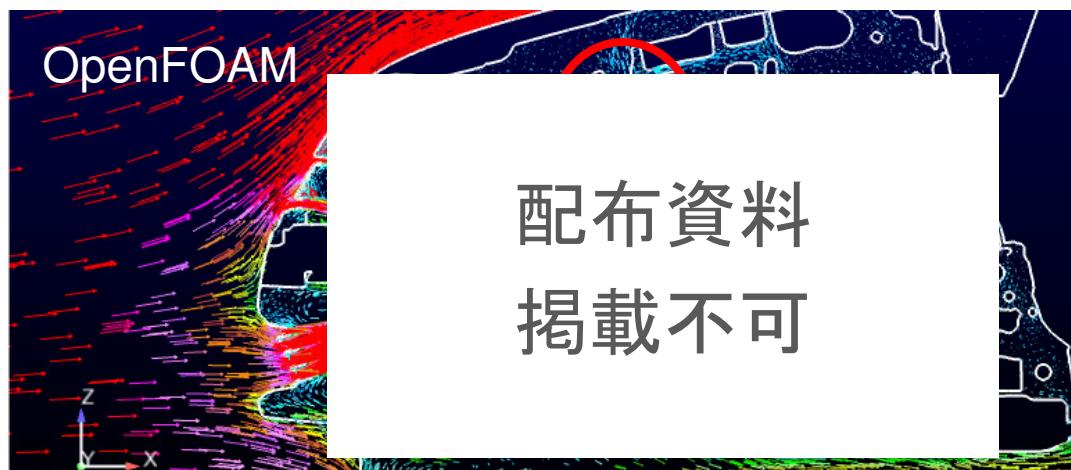
- Meshing done in 6.5 days using ICEMCFD and ANSA
- 10 prism layers with first cell height 1mm
- 29M cell count



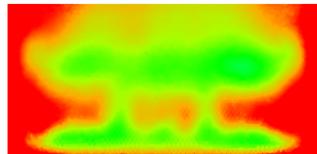
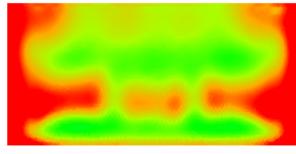
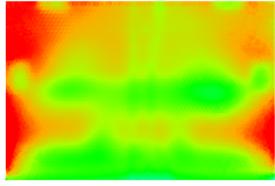
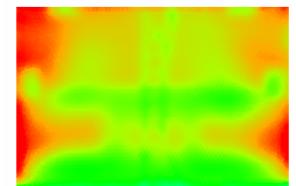
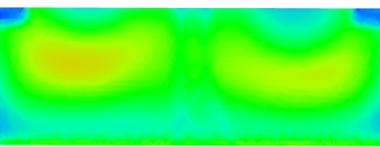
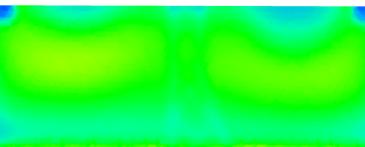
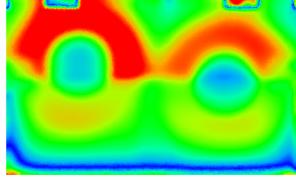
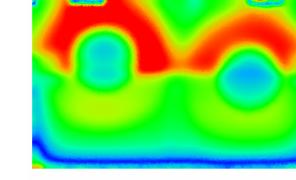
事例(コード比較) :Aerodynamic (Incompressible Turbulent Flow: Real Car)

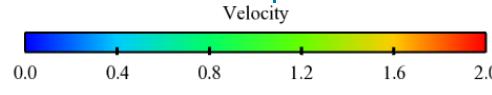


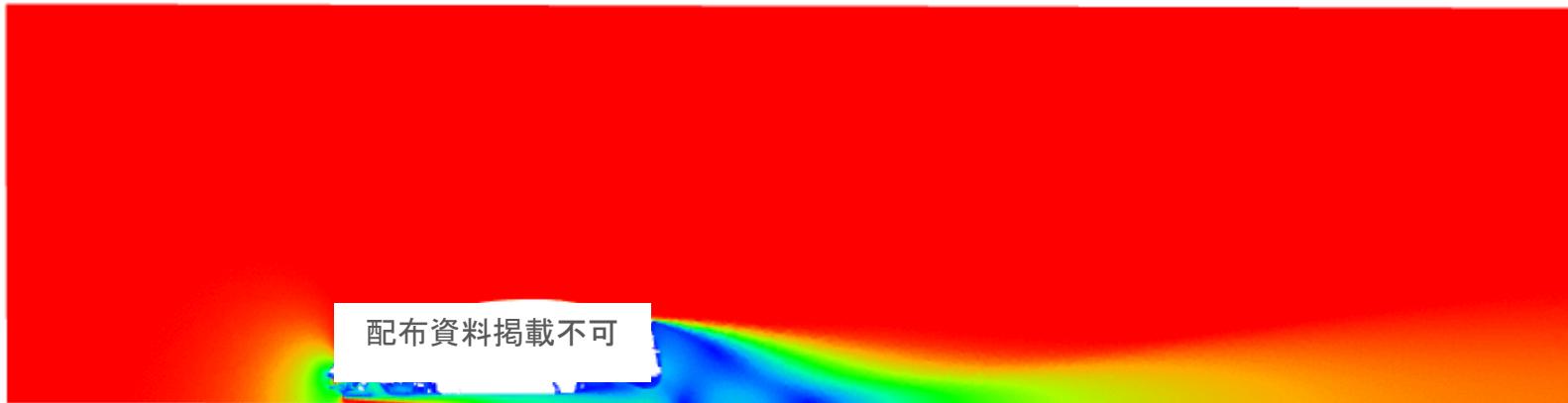
- Flow field comparisons were made between the commercial code and OpenFOAM
- OpenFOAM solution looks comparable
- Slight differences in the solution are a result of differences in the mesh used.



事例(コード比較) : Aerodynamic (HX Flow Rates for 50km/hr Vehicle Speed – No Airdam)

Flow through Heat Exchangers	OpenFOAM		Other Commercial Code	
	Flow Rate (m ³ /min)	Plot	Flow Rate (m ³ /min)	Plot
BAT	21.87		21.12	
CNDSR	23.39		22.58	
PE	6.35		6.11	
ATO	8.57		8.09	
RAD	18.08		17.63	





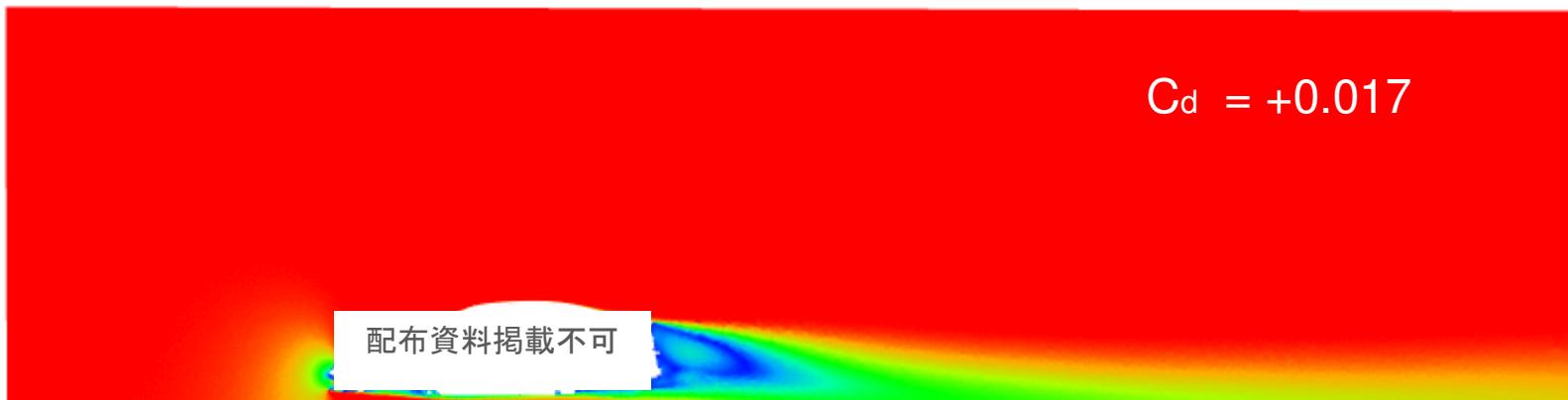
With Airdam

Velocity Magnitude



Results were close to Test and Other Commercial Code data

$$C_d = +0.017$$

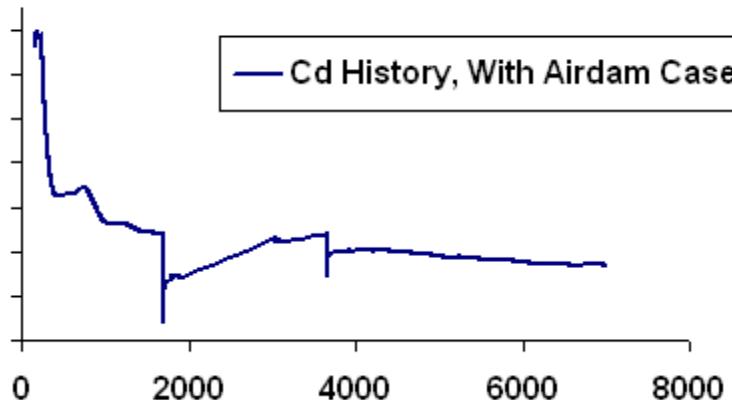


Without Airdam

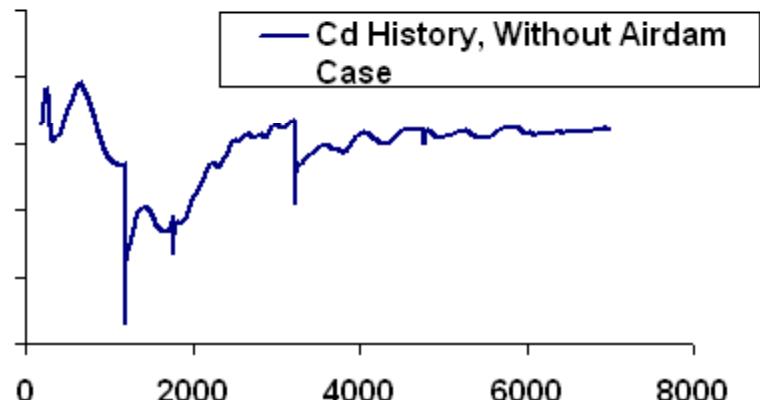
事例(コード比較) :Aerodynamic (Incompressible Turbulent Flow: Real Car)

OpenFOAM results were in-line with Test results

With Air dam



Without Air dam

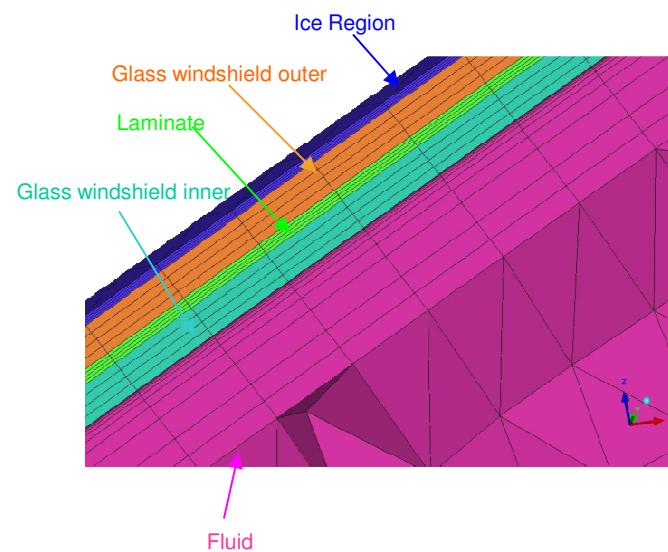
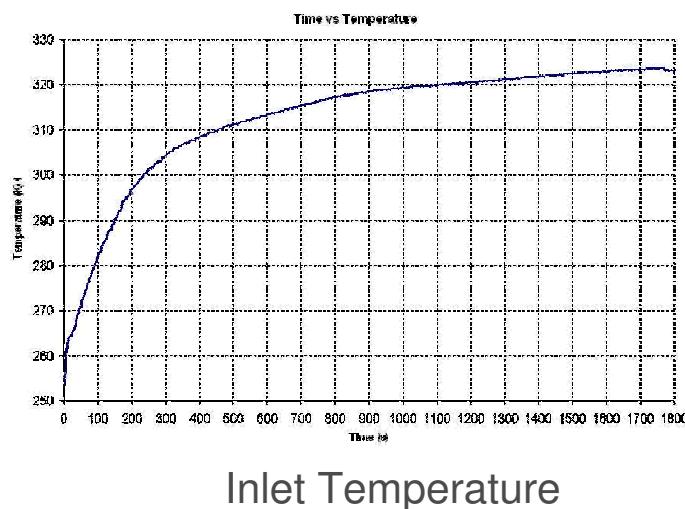


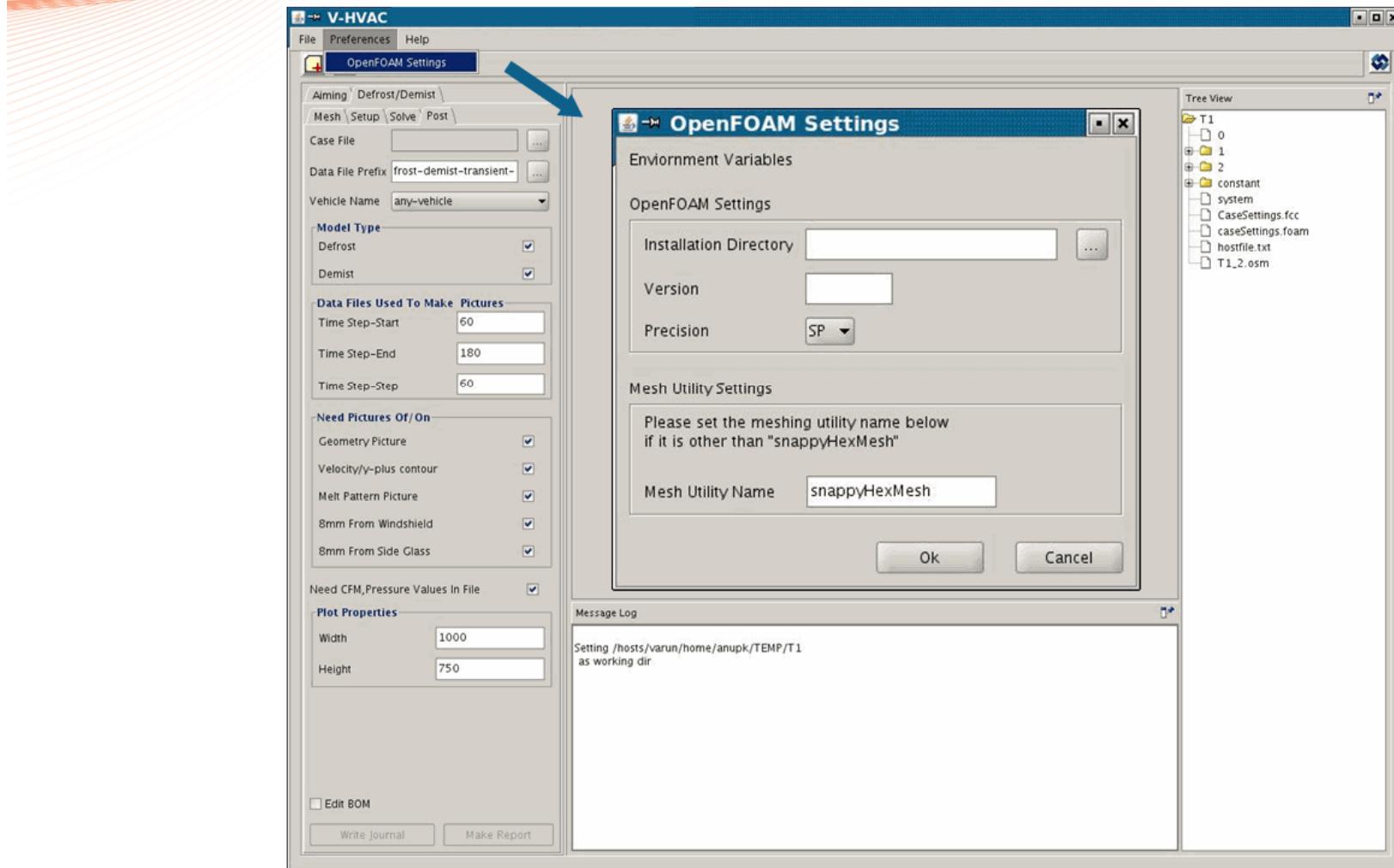
商用コードとの比較事例

Defrost

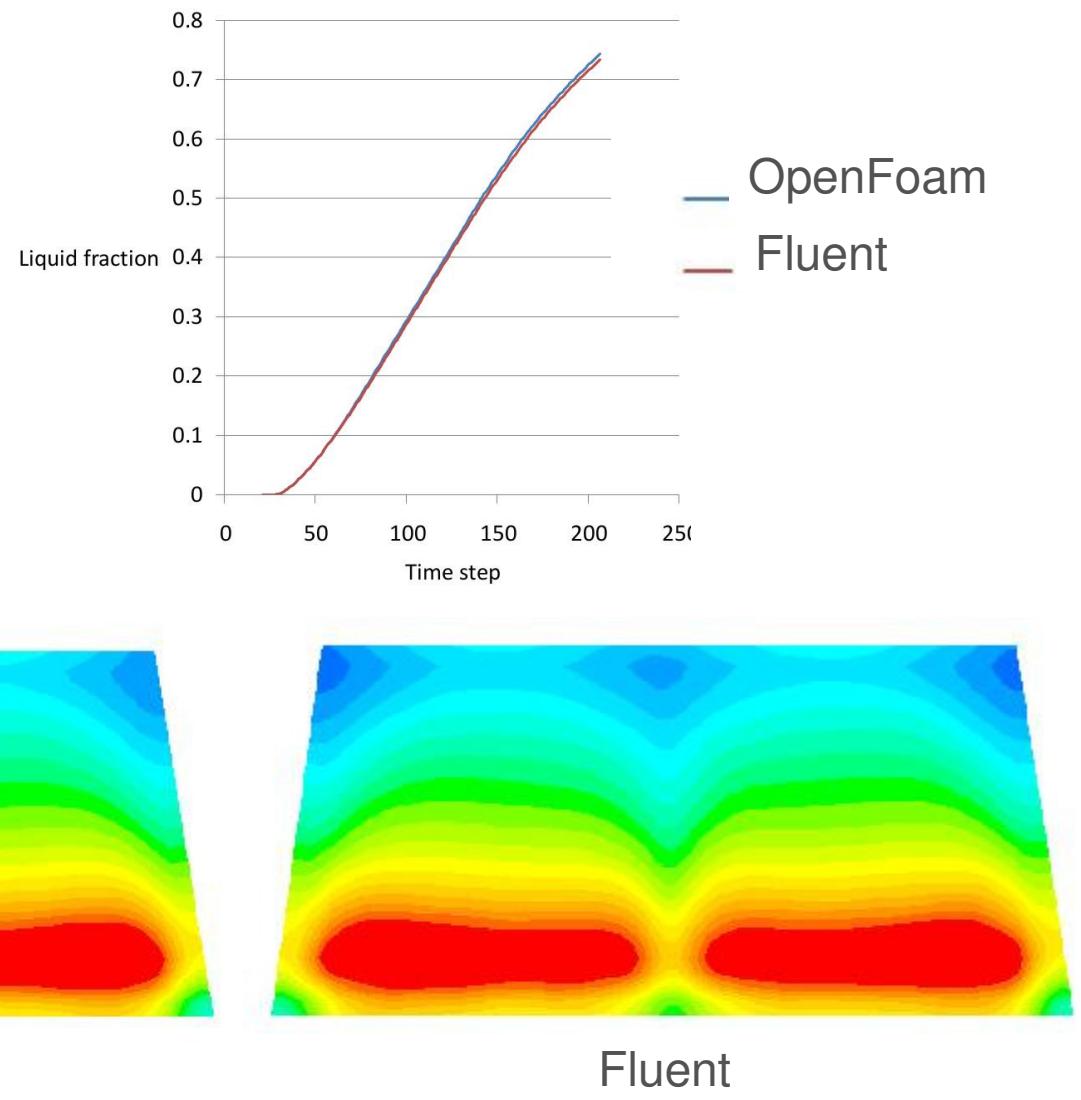
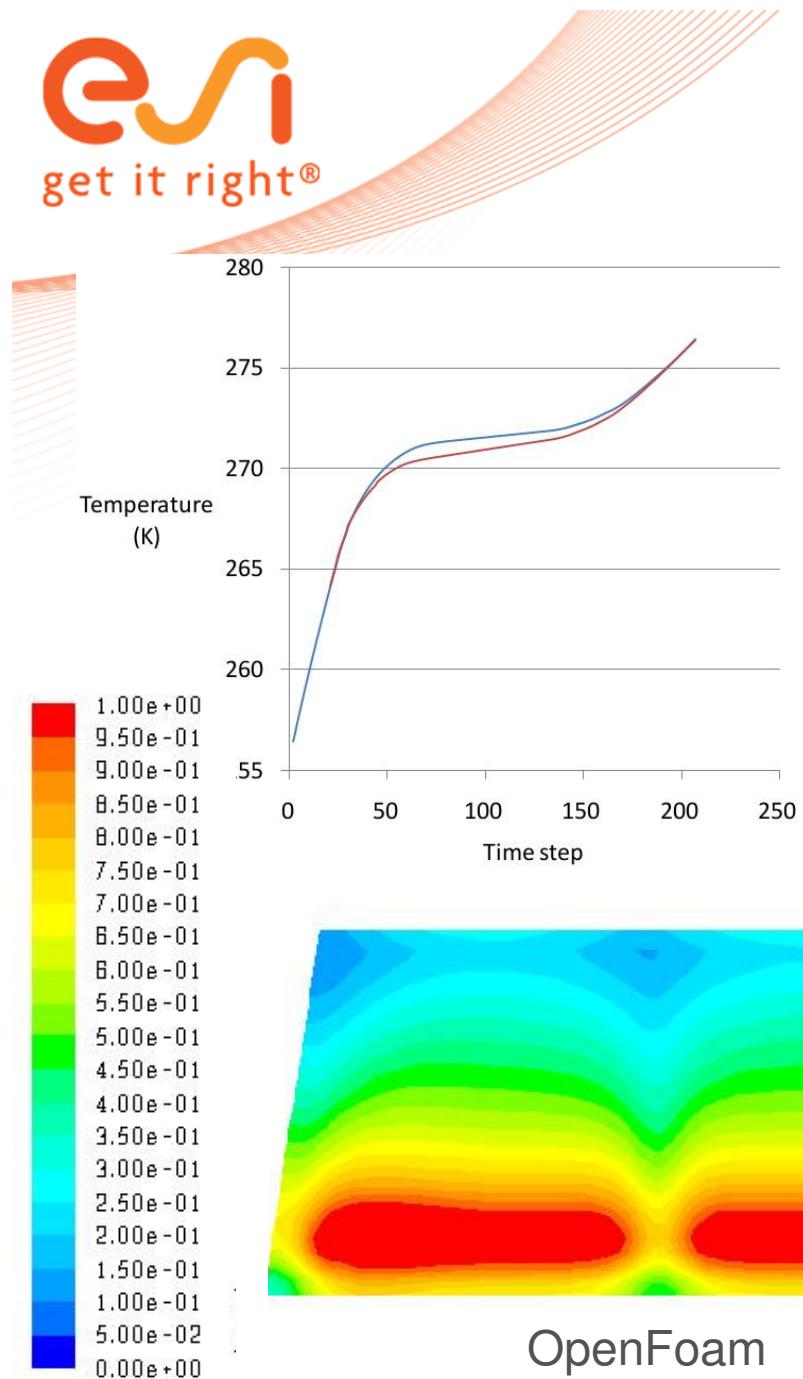
事例 (コード比較) : Defrost

- Melting of ICE
- Tetra/Prism Mesh (about 10.3M Cells)
- Velocity Inlet at Duct (3.776 m/s) and Pressure Outlet at back
- Turbulence Model: Realizable k-epsilon is used in Fluent and k-omega SST in OpenFoam
- Time Step size for Ice melting simulation = 5 sec.





事例 (コード比較) : Defrost





事例 (コード比較) : Defrost
Contours on Ice region top layer

Time = 18 min

Fluent



OpenFoam



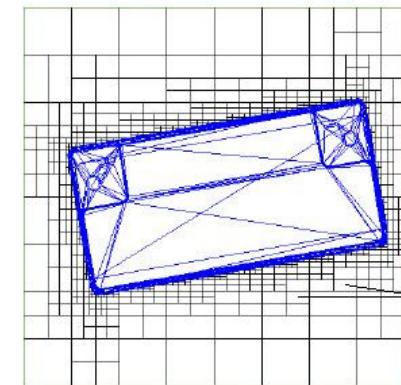
配布資料掲載不可



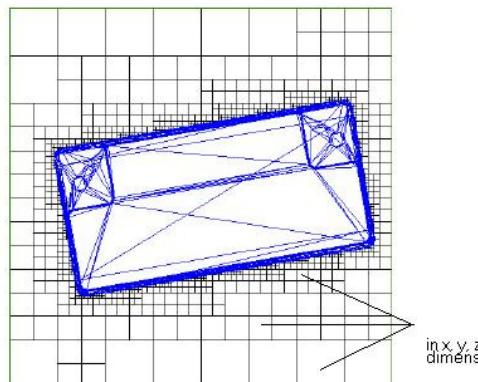
配布資料掲載不可

snappyHex vs. CFD-VisCART

ESI's PreProcessor



Omnitree Meshing



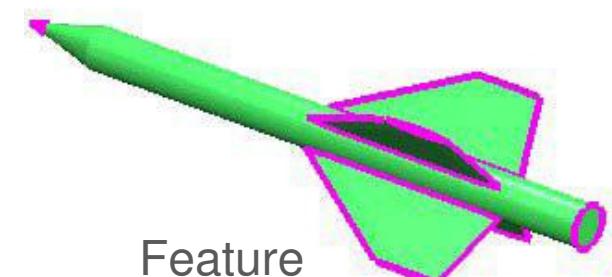
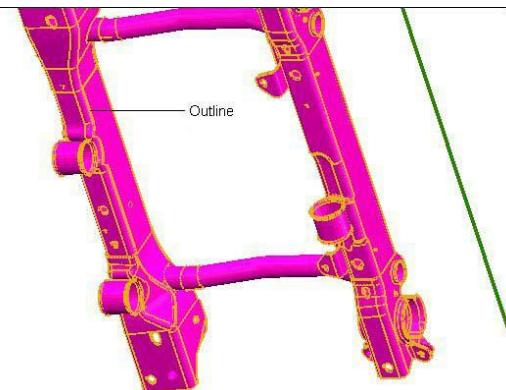
Octree Meshing

Leak Detection

CFD-
VisCART

Outline detection

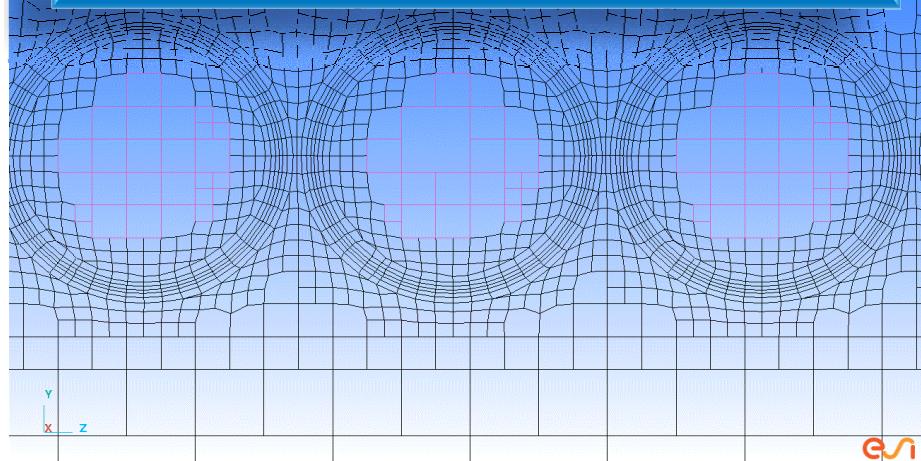
Mesh Generation
Stair-case
Single Domain
Multi-Domain
Shrink Wrap



Feature
extraction



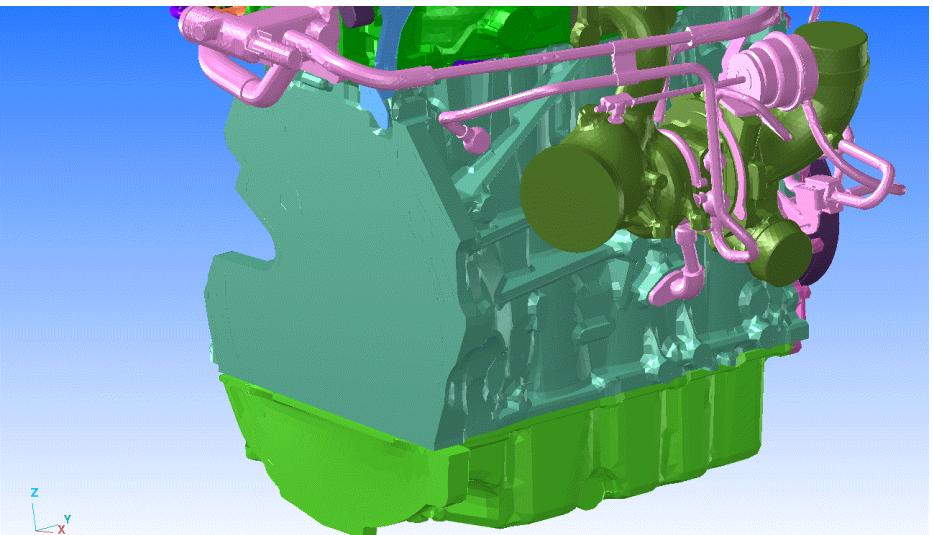
Quick, Powerful, Multi-Domain Meshing



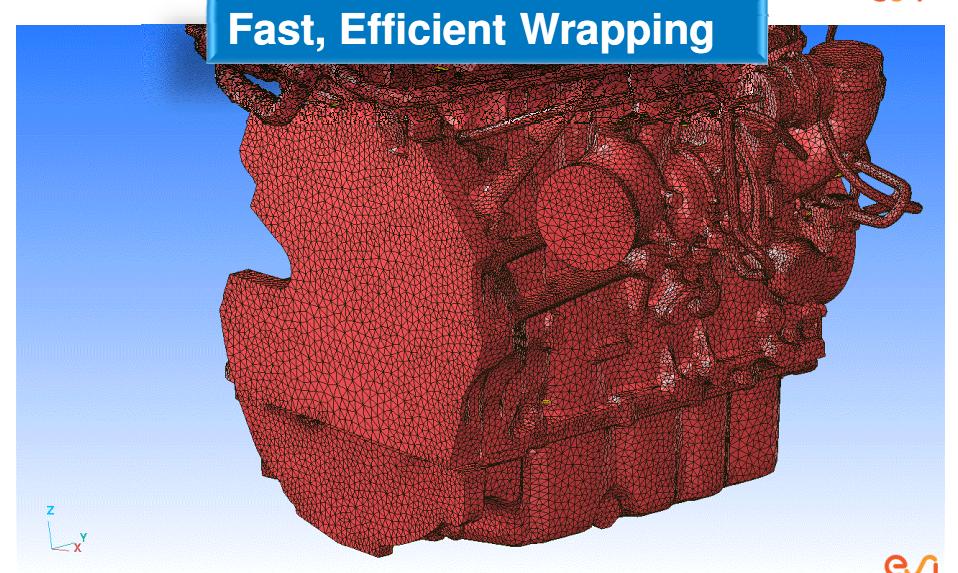
•Courtesy: Bissell Inc.

CFD-VisCART

Quick, yet powerful meshing

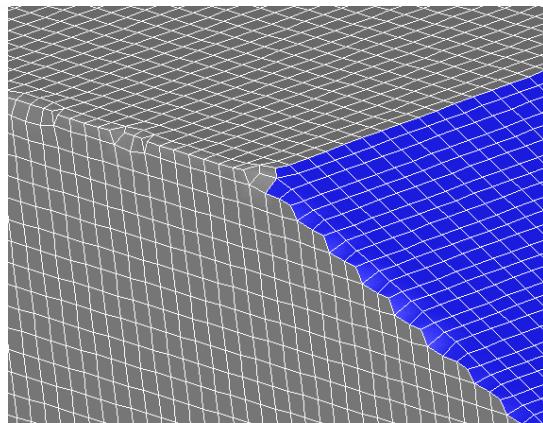


Fast, Efficient Wrapping

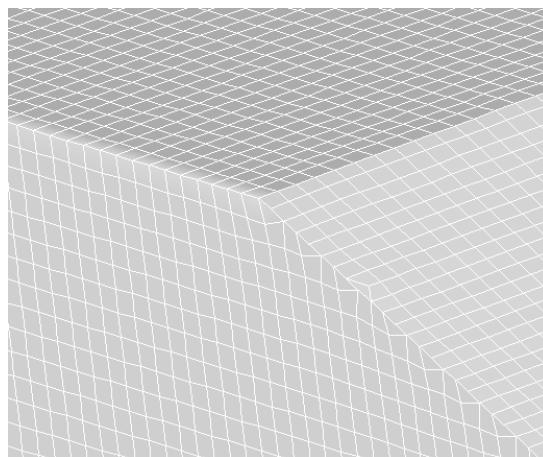


(Comparison for Ahmed Body: Feature Resolution, BL)

- SnappyHex does not resolve sharp feature angles (latest version does somewhat but not completely).

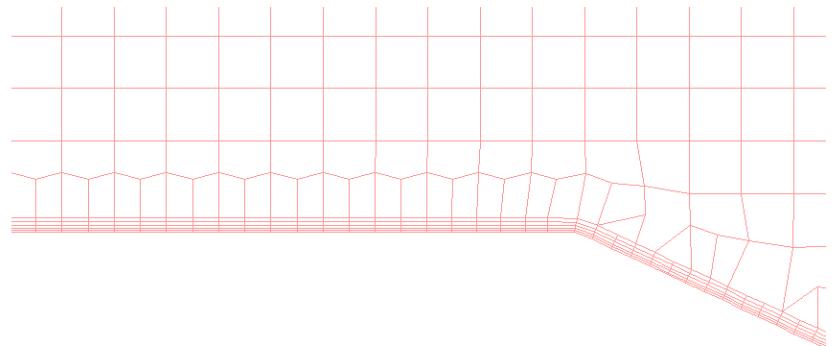
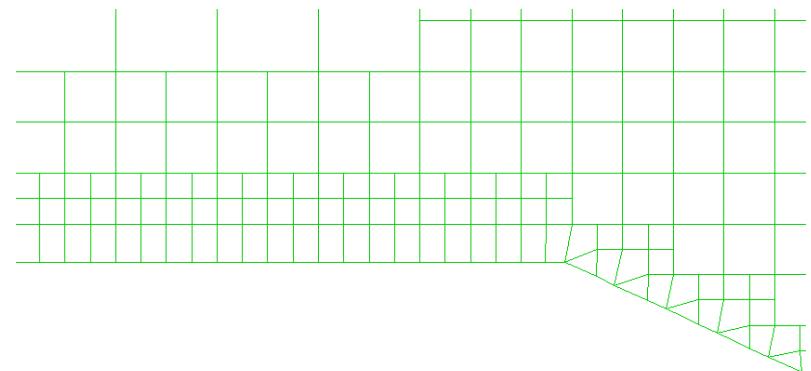


snappyHex



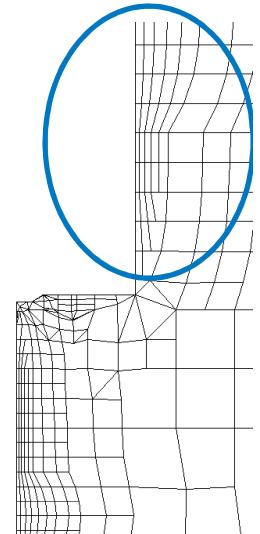
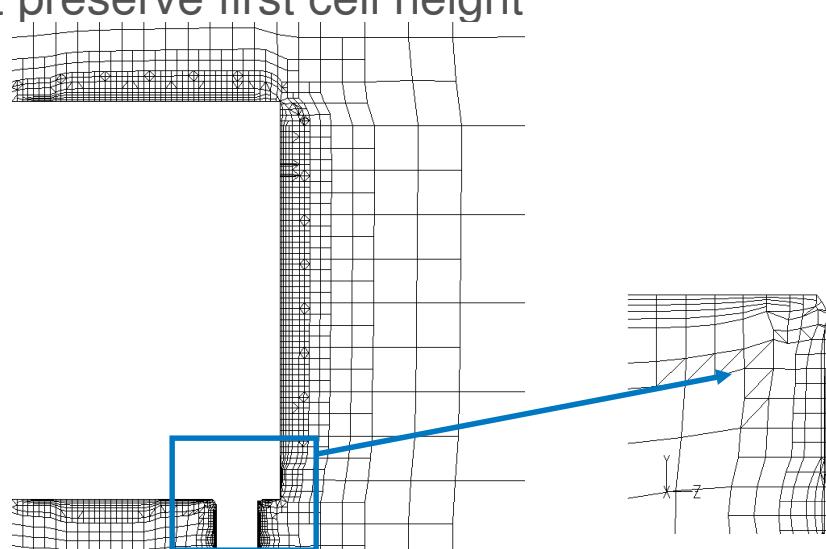
VisCART

- SnappyHex cannot show prism layers on symmetry plane.



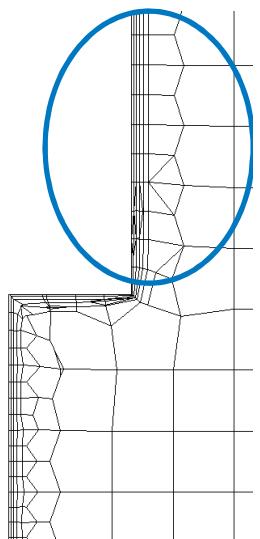
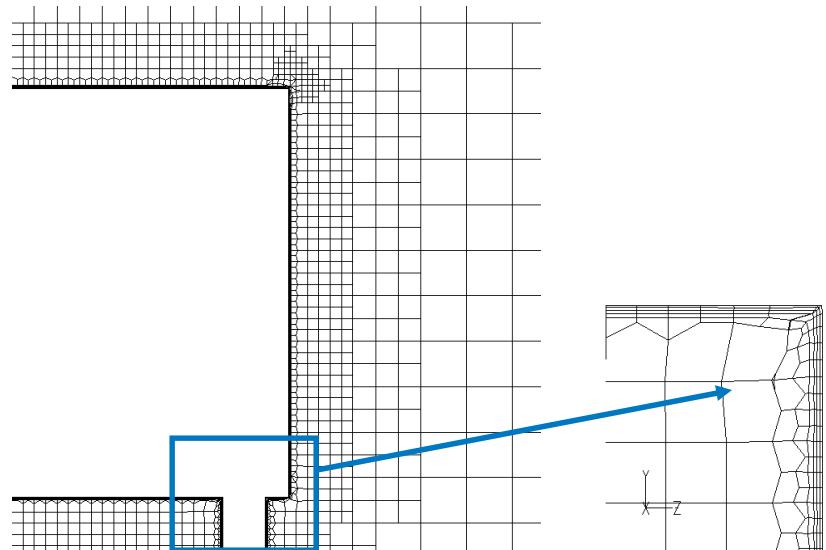
- SnappyHex does not preserve first cell height

snappyHex



- VisCART gives option to choose between Expanding or Fixed height.

VisCART



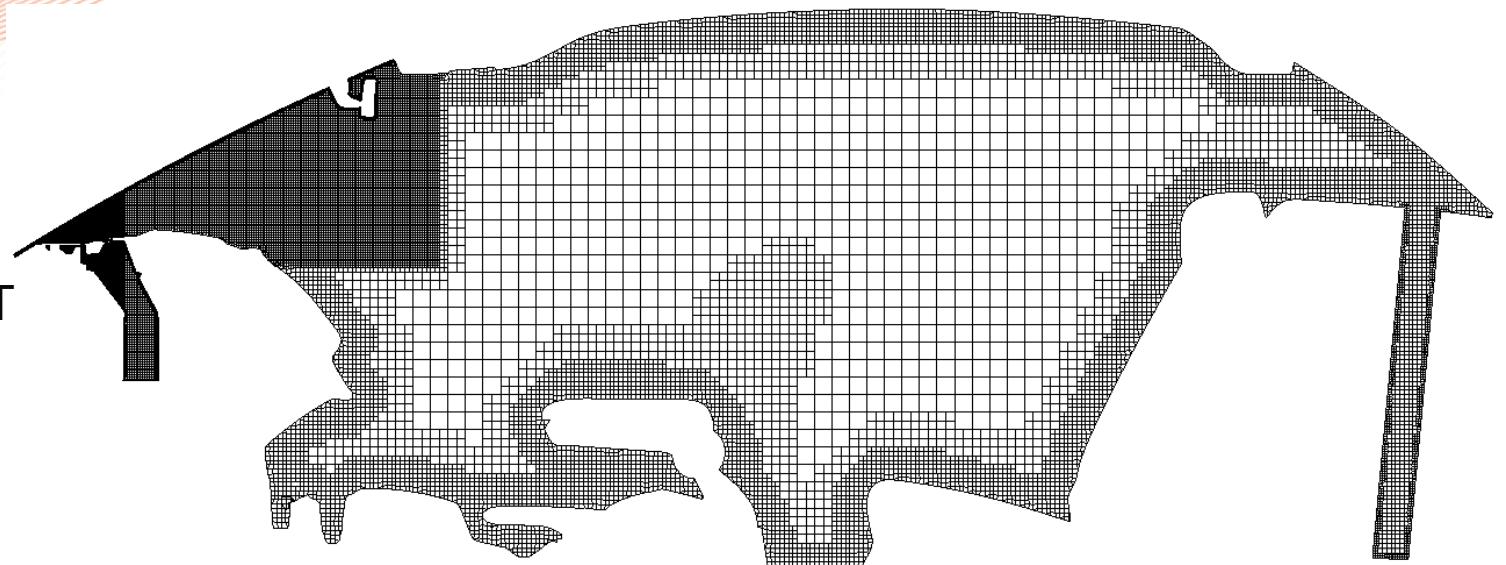
事例:(VisCART メッシュ適応) (Industrial Test Case: Automotive Defrost/Demist)

- Mesh has been created in CFD-VisCART (as well in Snappy)
- Surface sizes, Refinements, Boundary Layer Parameters kept exactly same as in snappyHexMesh
- Prism layer information
 - First Cell Height – 0.2 mm
 - Expansion Ratio – 1.3
 - Total No. of Layers - 6
- Total Mesh size 12.2 million

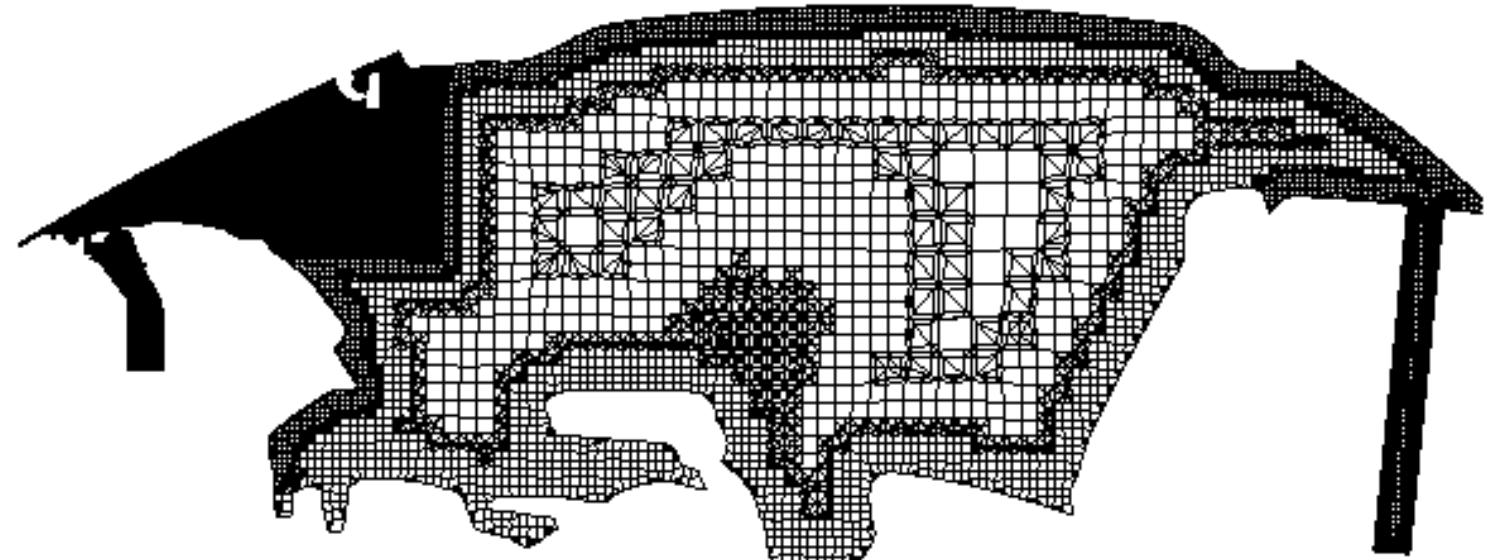


事例:(VisCART メッシュ適応) メッシュの比較

CFD-VisCART

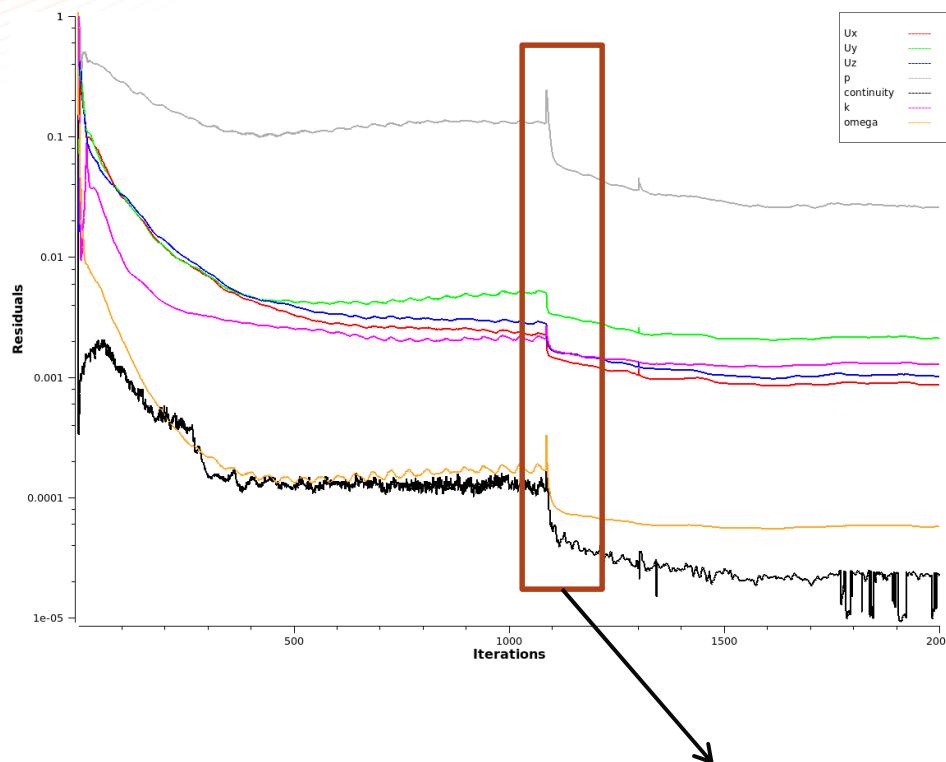


snappyHexMesh

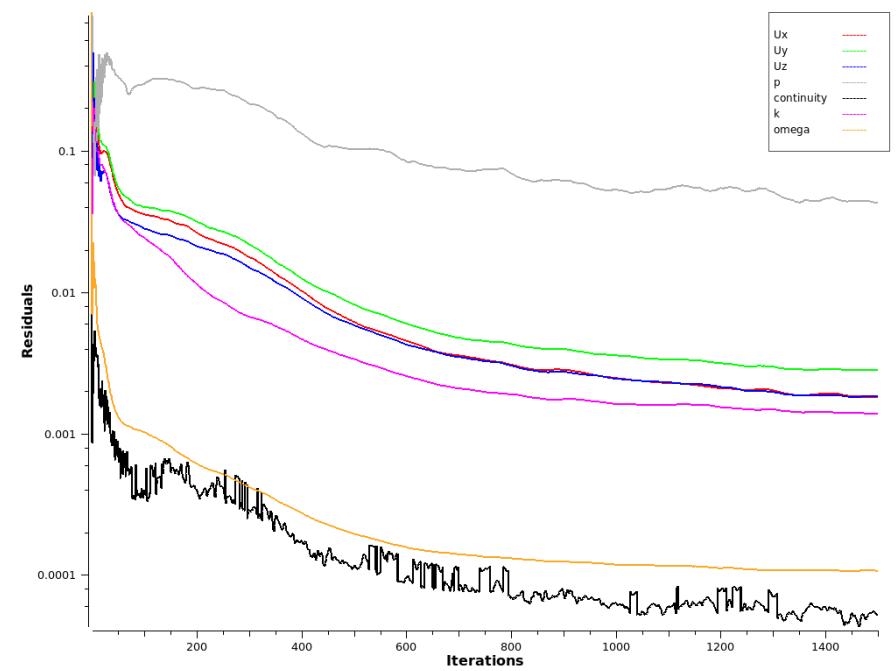


Results - Residual

CFD-VisCART



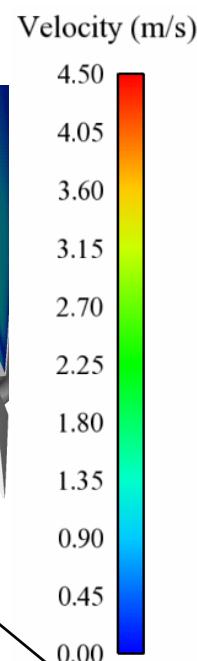
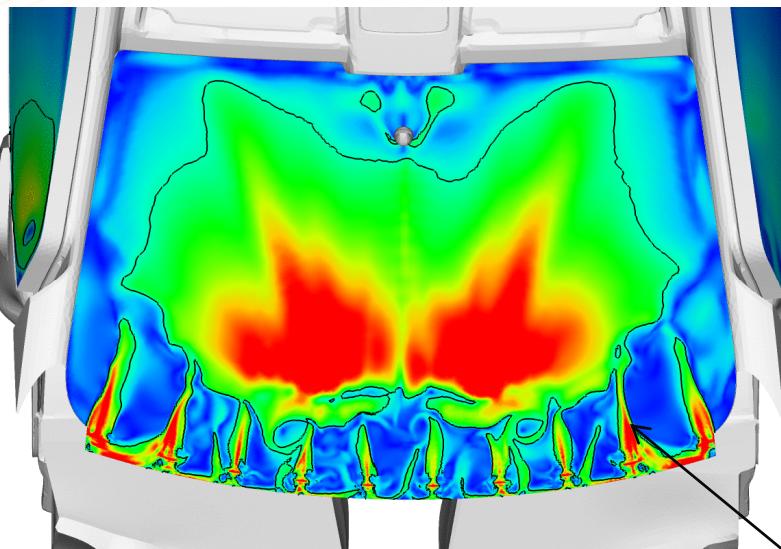
snappyHexMesh



This is because of reduction
in relaxation parameters.

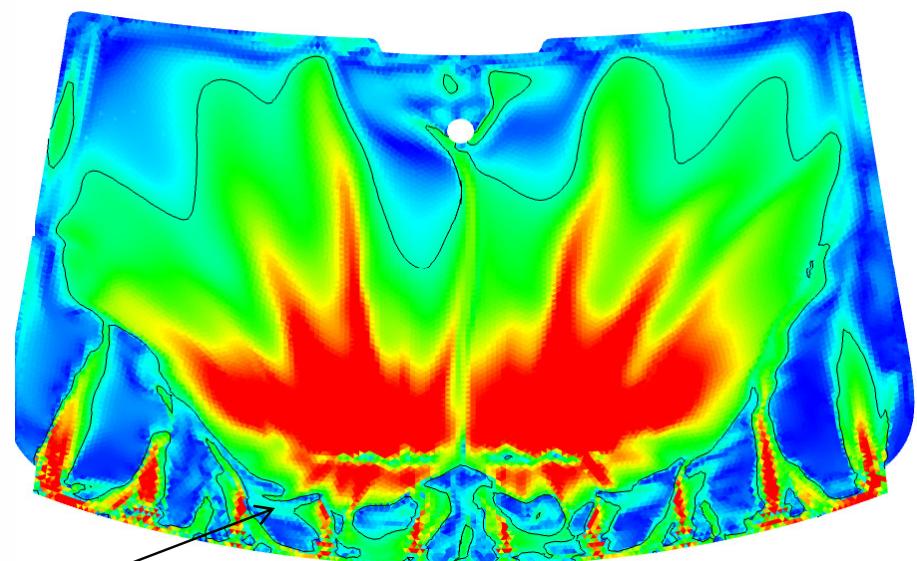
Velocity Contour on 2.5 mm away from Windshield Glass Wall

CFD-VisCART

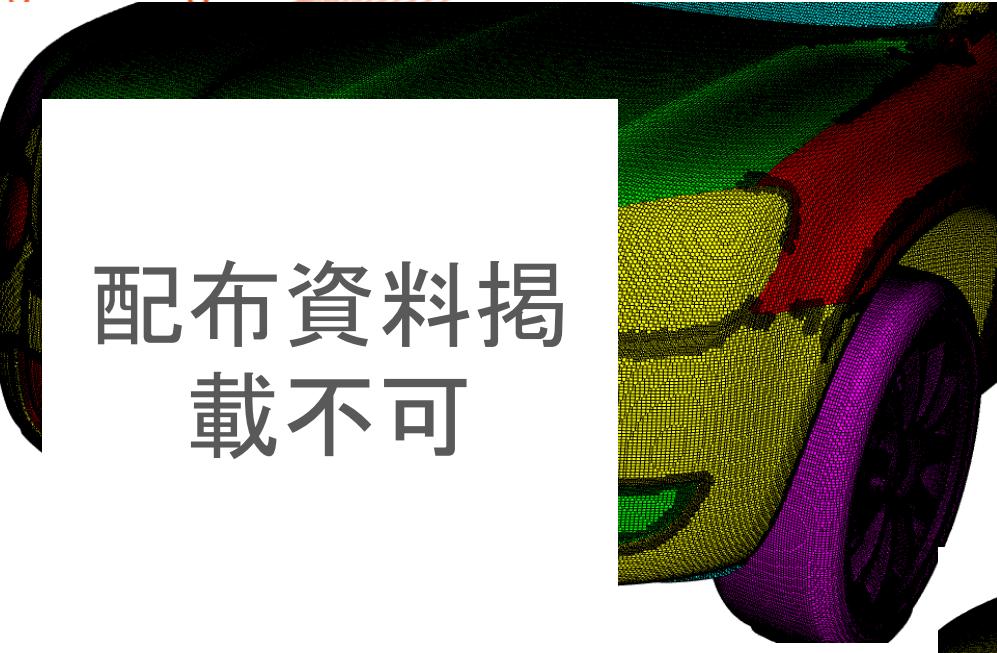


Isoline at 1.5 m/s

snappyHexMesh



The spike in Velocity Contour on the RHS is not physical.



CFD-VisCART

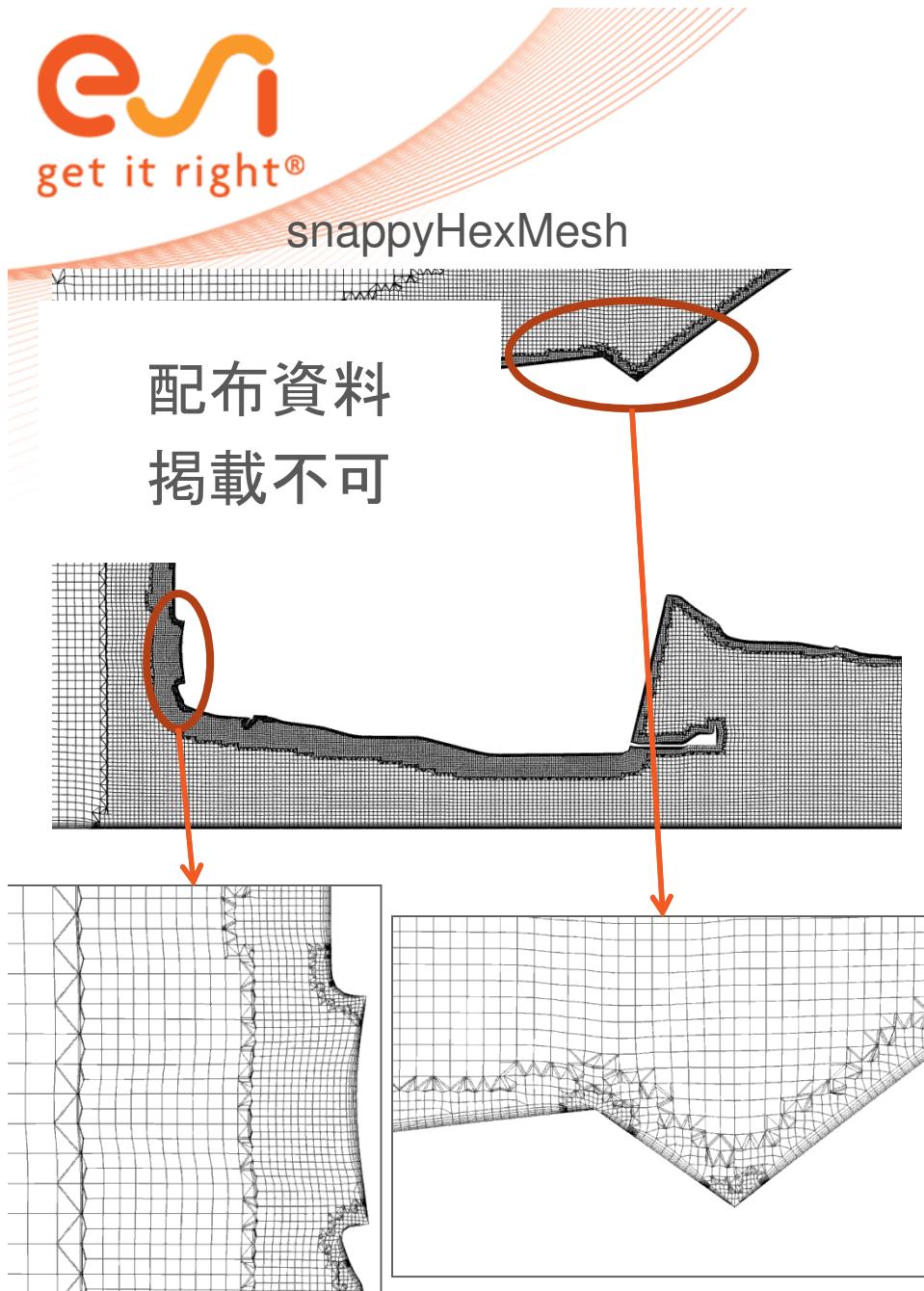
ESI's preprocessor
Mesh size: 30.7 Million

事例:(VisCART メッシュ適応)
(Mesh Comparison)

snappyHexMesh

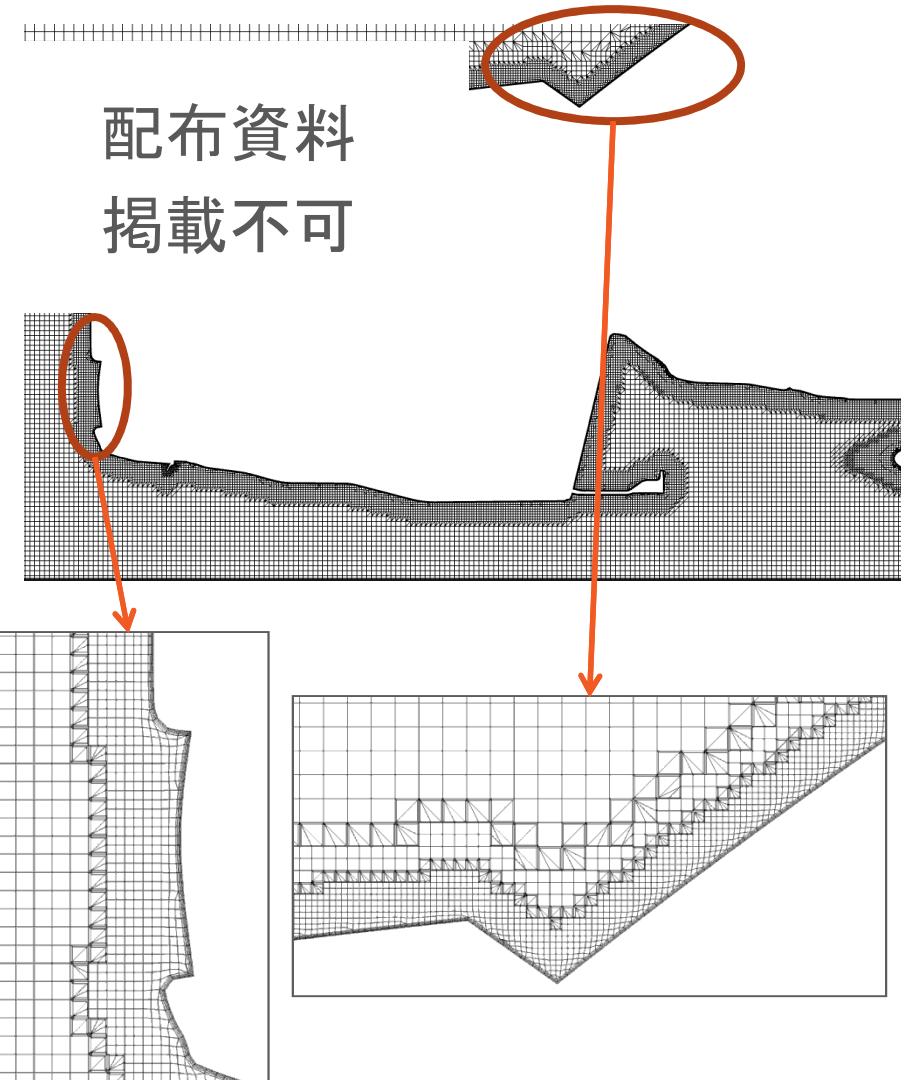
OEM's own version of snappyHex
Mesh size: 45.4 Million



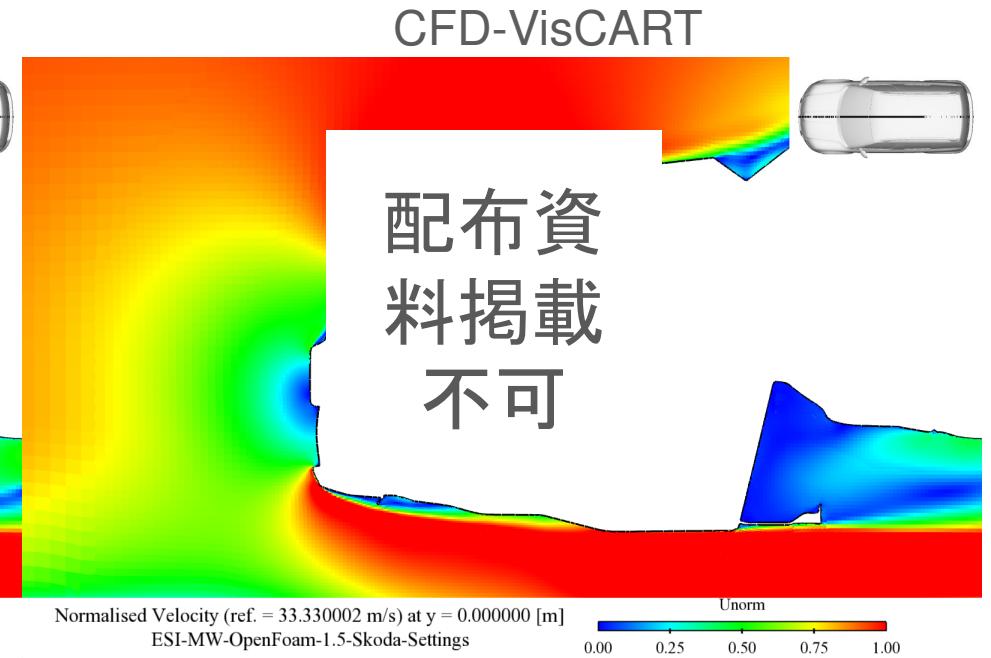
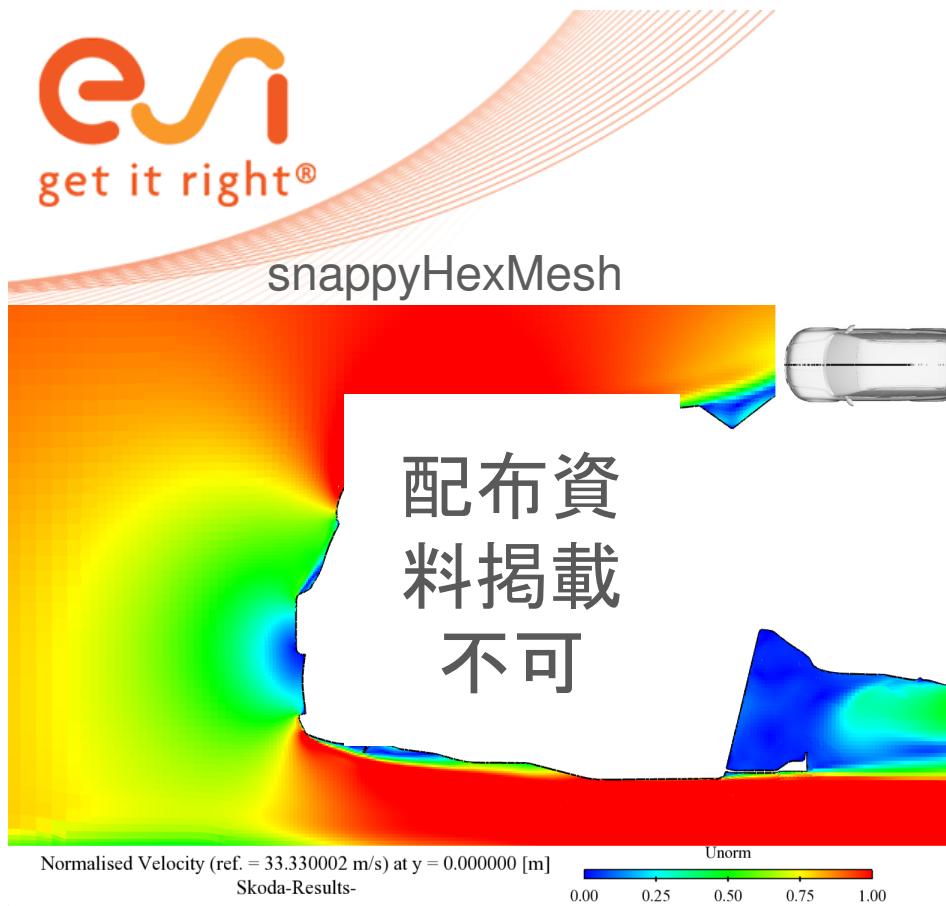


事例:(VisCART メッシュ適応) (Mesh Comparison)

CFD-VisCART



事例:(VisCART メッシュ適応) Domain Plots Velocity (normalized)



Customer quite satisfied, next phase to start!

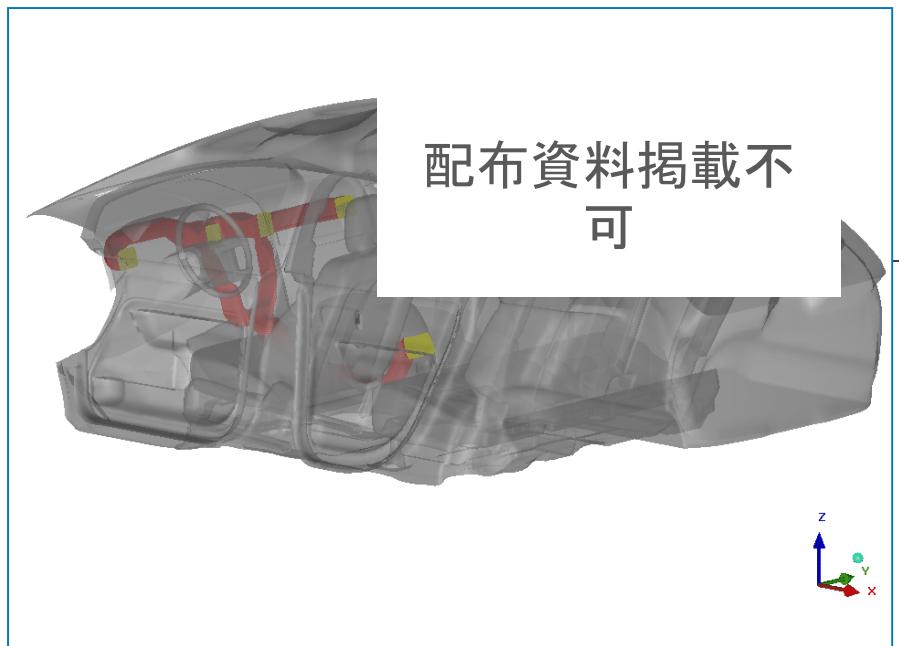
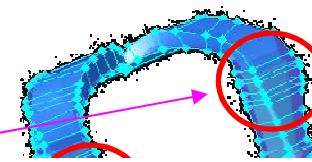
OF Version	Mesh Tool	Cd	CSF (3%)	Cl (total)	Cl front	KFU (0.040)	Cl rear	CSF (0.010)	Remarks
Experiment	-	0.353	0.011	0.085	-0.049	0.040	0.134	0.010	OEM Measurement
OEM (1.5-1)	snappyHexMesh	0.351	0.002	0.091	-0.043	0.006	0.134	0.000	OEM Result
ESI-OF-1.5	ESI-VisCART	0.353	0.000	0.087	-0.047	0.002	0.135	0.001	Ran on ESI Cluster



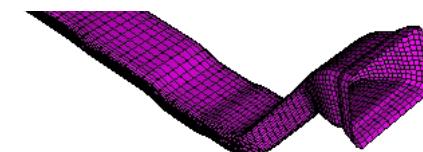
他ツールとの連携 最適化ソフト

事例:Duct Optimization Morphing using Sculptor

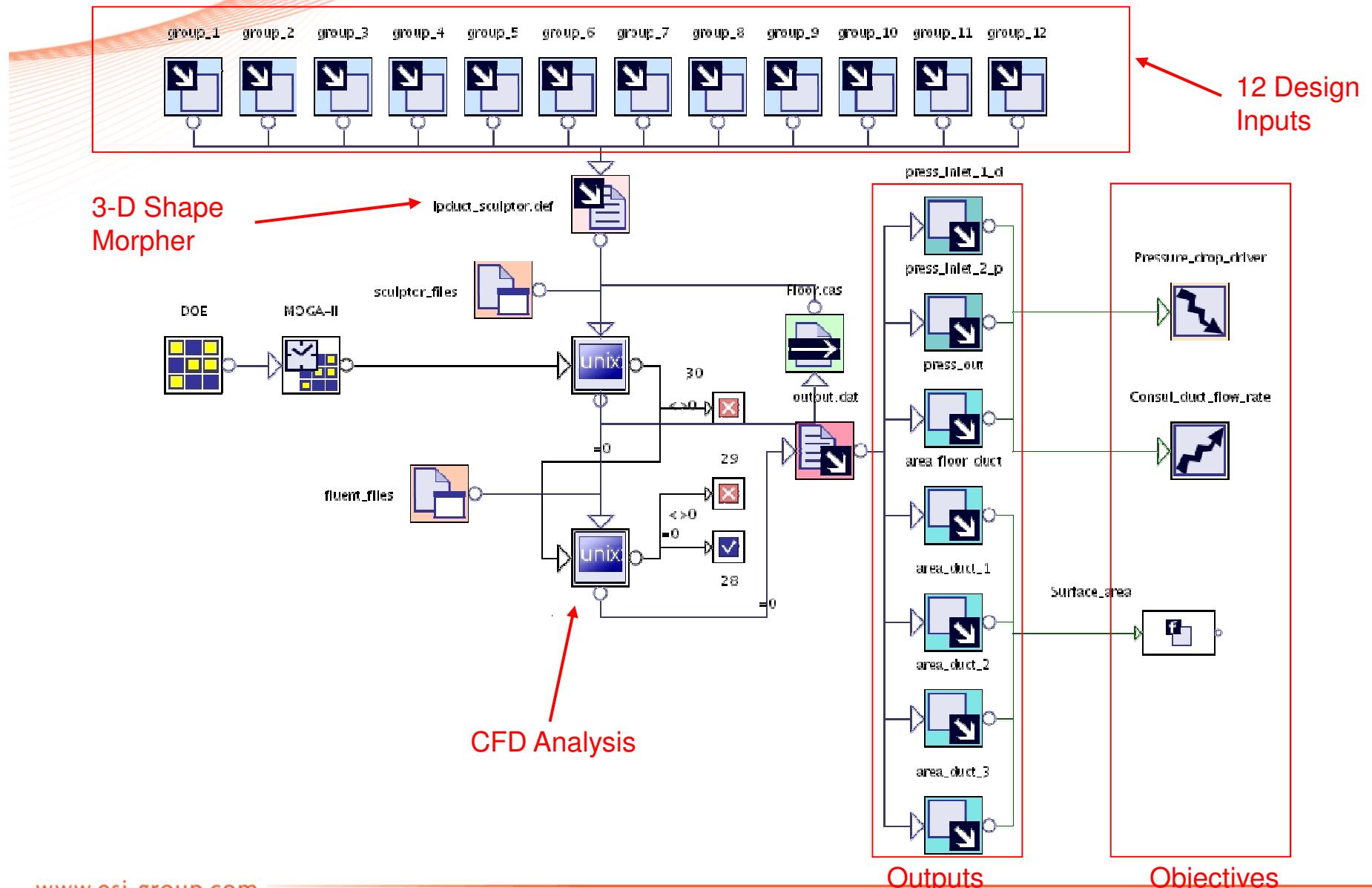
Control Group
Locations for
mesh morphing



配布資料掲載不可



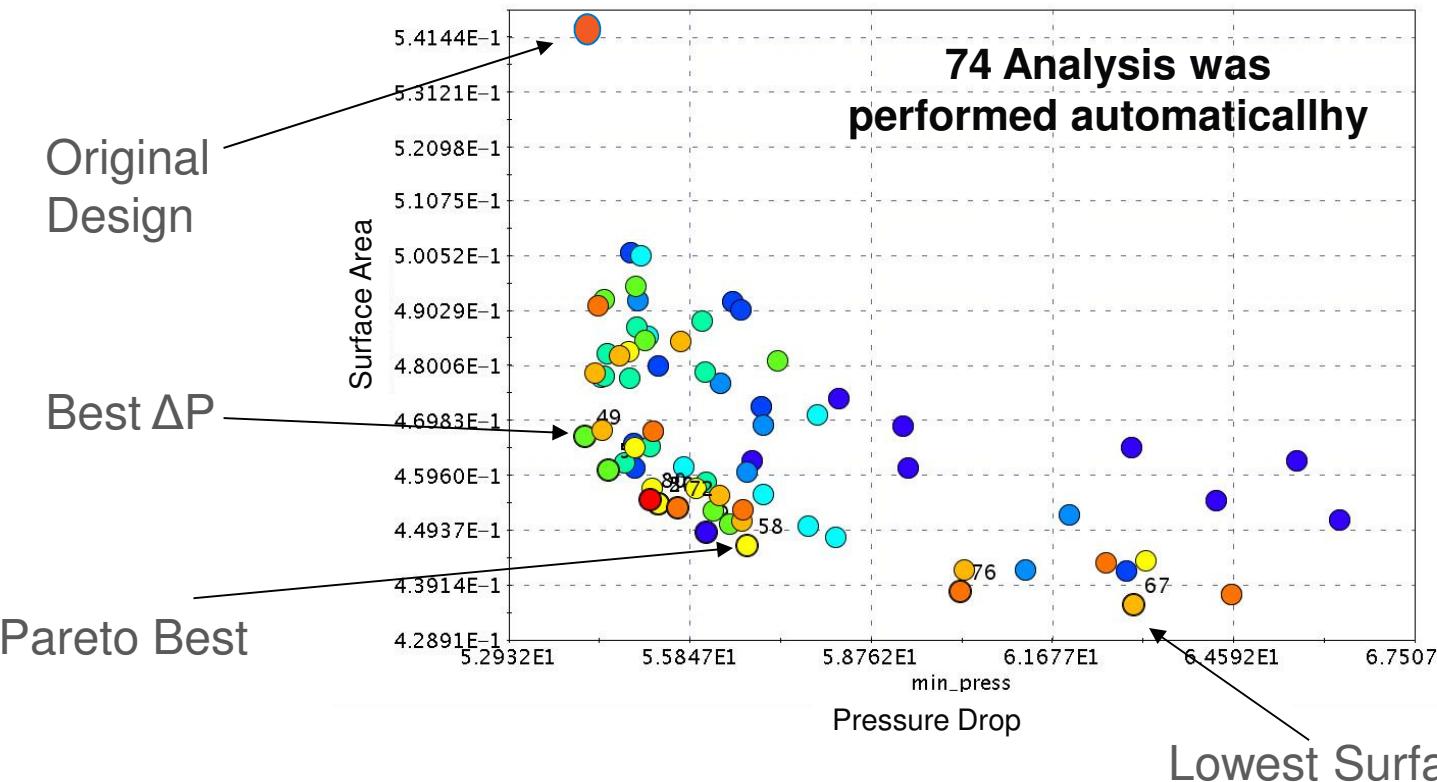
Optimization Process

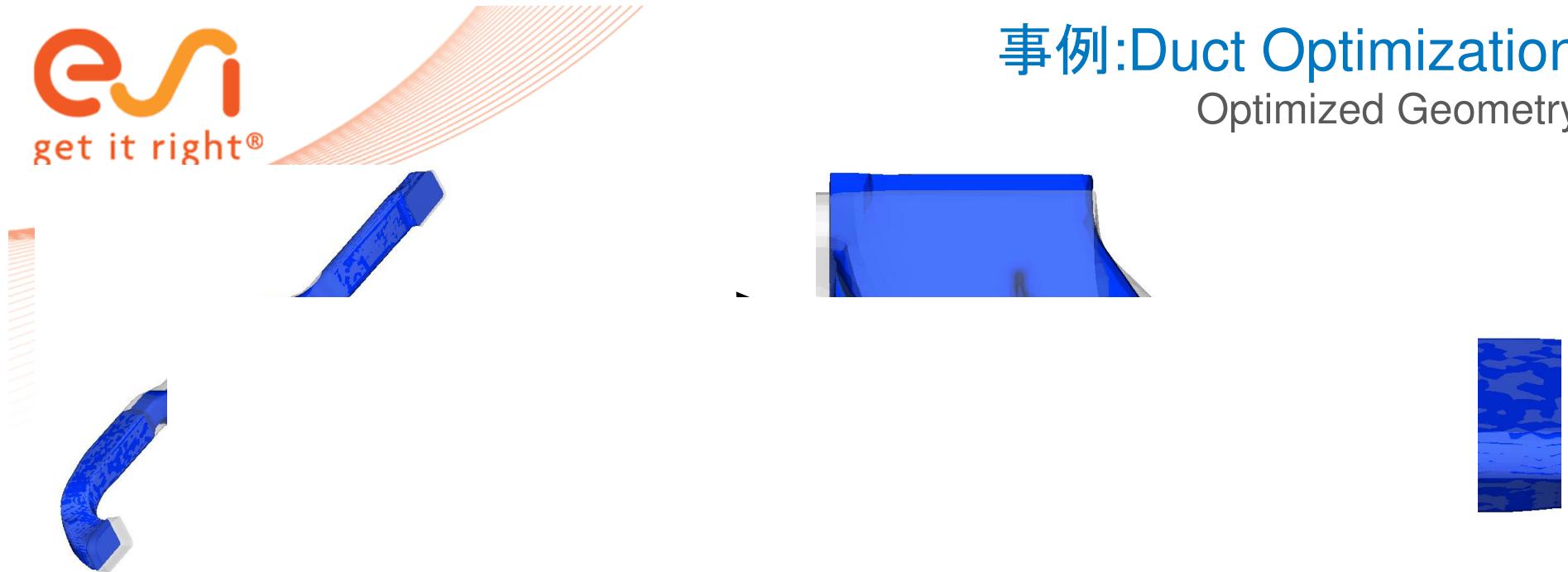


Plot of Pressure Drop (Horizontal axis)

vs.

Surface Area (Vertical Axis)





配布資料揭載不可



Grey: Baseline

Blue: Optimized





事例:Duct Optimization

Comparison of Baseline Vs Optimized Design

- Reduced testing and elimination of prototype tooling costs
- \$150,000 Savings/Vehicle Program
- ~\$750,000 / Year

VET OCTIV m/s

配布資料揭載不可

	Outlet Driver Side	Outlet Panel Driver Side	Outlet Panel Passenger Side	Outlet Passenger Side	Outlet Console
Mass Flow (%) (Baseline)	23.7	20.9	24.6	23.2	7.6
Mass Flow (%) (Optimized Shape)	21.4	21.2	20.1	20.2	17.1



www.esi-group.com